Fairlane Wheel Centrifugal Stresses

The Team 100 shooter uses Fairlane neoprene rubber wheels, McMaster-Carr part number 2476K37. There is a question whether the wheels can be spun at the speeds necessary to launch power cells (foam balls) in the 2020 Infinite Recharge FRC game without over-stressing the wheels. Accordingly, we have performed a axisymmetric stress analysis to calculate the radial and hoop stresses and the radial displacements in the rubber and steel. The wheels are 4-inch-diameter by 2-inch wide with a tubular steel core having 1.25-inch ID and 1/16-inch wall thickness. The Durometer is 60A. In the analysis, Material 1 is designated by subscript 1 and represents the steel core. Material 2 is the rubber.



Made of neoprene rubber, these rollers resist oil, flames, gasoline, and weather. Mount them directly onto your shaft or build a custom roller by adding your own hub.

Jurometer (Hardness Rating): 60A (Medium) Black

Neoprene (chloroprene) Durometer 60A rubber properties are taken from the MatWeb page http://www.matweb.com/search/DataSheet.aspx?MatGUID=c680960fc38f488a991e1e04504092b4. The relevant properties for the stress analysis are the modulus of elasticity (Young's modulus) and density (specific gravity).

Categories: Polymer, Thermoset; Rubber or Them	ioset Elastomer (TSE)		
Material Notes: UL-flame resistant, non-blooming neop	rene Inc		
Key Words: Polychloroorane Rubber (CR)			
Vendere: No update are listed for this material	Diagra aliak hara ifugu ar	n a supplier and would like	information on how to add your listing to this material
vendors. No vendors are listed for this material.	riease <u>click fiere</u> li you a	e a supplier and would like	information on now to add your listing to this material.
Physical Properties	Metric	English	Comments
Specific Gravity	1.52 g/cc	1.52 g/cc	ASTM D297, 15
Machanian Descention	Matria	EK-b	Commute
Mechanical Properties	Metric	English	Comments
Hardness, Shore A	2= 08	80 =<	ASTM D2240, Type A
	72	72	After Dev Heat Asing: ASTM D888
	@Temperature 100 °C,	@Temperature 212 "F,	Alter bry heat Aging, ASTM boos
	Time 605000 sec	Time 168 hour	
Tensile Strength at Break	>= 11.88 MPa	>= 1723 psi	ASTM D412 Method A
	14.84 MPa	2152 psi	AVG; ASTM D412 Method A
	12.47 MPa	1808 psi	After Dry Heat Aging; ASTM D865
	@Temperature 100 °C,	@Temperature 212 "F,	
Tanala Olasa ata Viata	Time 605000 sec	Time 168 hour	AVO Observation MOS. AOTH D410 Mathed 4
Tensile Strength, Yiela	1.00 MPa	140 psi	AVG Stress at Elongation M20; ASTM D412 Method A Stress at Elongation M50; ASTM D412 Method A
	2~ 1.03 MPa	218 psi	AVC Stress at Elementian M50; ASTM D412 Method A
	1.48 MFa	210 psi	Avid Stress at Elegentian M100, ASTM D412 Method A
	2= 1./4 MPa 2.82 MPa	202 psi	AVG Stress at Elongation M100; ASTM D412 Method A
Elongation at Break	2.02 MFa	>= 297 %	AVG Stress at Elongation Mildu, ASTM D412 Method /
	382 %	382 %	AVG: ASTM D412 Method A
	188.5 %	188.5 %	After Dry Heat Aging: ASTM D88
	@Temperature 100 °C,	@Temperature 212 "F,	And Dry Heat Aging, Abrin book
	Time 605000 sec	Time 168 hour	
Modulus of Elasticity	0.00614 GPa	0.890 ksi	ASTM D412 Method A
Shear Modulus	0.00205 GPa	0.297 ksi	ASTM D412 Method A
Secant Modulus	0.00539 GPa	0.782 ksi	ASTM D412 Method A
Tear Strength Test	198	198	(psi); ASTM D624, Die E
Compression Set	17 %	17 %	ASTM D395, Test E
	@Temperature 100 °C,	@Temperature 212 *F,	
	Time 252000 sec	Time 70.0 nour	
Thermal Properties	Metric	English	Comments
Glass Transition Temp, Tg	-40.0 °C	-40.0 °F	ASTM D132
Descriptive Properties			
Strain Energy / Unit Volume at 20% Elongation (psi)		16	

since on the valued value provided in the property value to see the original wave and uncorrections but to display the indianation in a contraction where original value are a source or significant and contractions can click on the property value to see the original value see and as a source or indianation in a contraction where the original value or early first and contractions to minimize rounding error. We also ask that you refer to MatWeb's terms of use parving this information. <u>Click here</u> to view all the property values for this dataset as they were originally entered into adVeb.

The "Tensile Strength, Yield" values given above are not yield stresses in the traditional sense of the onset of permanent (plastic) deformation, but rather stresses at 25%, 50%, and 100% elongations as means for characterizing the nonlinear stress-strain curve at large strains. As will be shown, predicted strains at 10000 RPM are around 27% maximum. See https://moldeddimensions.com/blog/defining-tensile-strength-elongation-and-modulus-for-rubber-and-cast-polyurethane-materials/.

The analysis solves for the stresses and radial displacement as a function of the radius. The analysis is well known, and is typically used to evaluate spinning rotors, thick-walled pressure vessels, and interference fits.



Parameters

Parameters used in the analysis are: a = inside radius = 0.625 in = 15.9 mm b = radius at interface between steel and rubber = 0.6875 in = 17.5 mm c = outside radius = 2.0 in = 51 mm $\sigma rr, \epsilon rr$ = radial stress and strain components $\sigma \theta \theta, \epsilon \theta \theta$ = hoop stress and strain components $\sigma zz, \epsilon zz$ = axial stress and strain components (σzz = 0 in the assumed case of plane stress applicable to a relatively thin disk, and ϵzz is not evaluated, although it could be.) v1 = Poisson's ratio of steel = 0.3 v2 = Poisson's ratio of rubber = 0.5 (I.e., in usual fashion, rubber is assumed incompressible. That means the shear modulus is significantly less than the bulk modulus) e1 = Young's modulus of steel = 209 GPa e2 = Young's modulus of 60A Durometer rubber = 6.14 MPa $\rho 1$ = density of steel = 8000 kg/m³ $\rho 2$ = density of polychloroprene rubber = 1520 kg/m³

Equations

The equilibrium equations for the two materials in terms of stress are:

$$\ln[1]:= \ deqs = \left\{ \\ \sigma rr1'[r] + \frac{1}{r} (\sigma rr1[r] - \sigma \theta \theta 1[r]) + \rho 1 \omega^2 r = 0, \\ \sigma rr2'[r] + \frac{1}{r} (\sigma rr2[r] - \sigma \theta \theta 2[r]) + \rho 2 \omega^2 r = 0 \\ \right\};$$

(See R. Budynas, *Advanced Strength and Applied Stress Analysis* (1977, McGraw-Hill, p. 139.) The strain-displacement compatibility equations for the two materials are:

Compatibility:

$$\frac{\partial^2 \epsilon_{\theta}}{\partial r^2} + \frac{1}{r^2} \frac{\partial^2 \epsilon_r}{\partial \theta^2} + \frac{2}{r} \frac{\partial \epsilon_{\theta}}{\partial r} - \frac{1}{r} \frac{\partial \epsilon_r}{\partial r} = \frac{1}{r} \frac{\partial^2 \gamma_{r\theta}}{\partial r \partial \theta} + \frac{1}{r^2} \frac{\delta^2 \epsilon_r}{\partial r^2} + \frac{1}{r^2} \frac{\delta^2 \epsilon_r$$

(See http://www.ecs.umass.edu/~arwade/courses/str-mech/polar.pdf)

The compatibility equations are second order differential equations in terms of strain or stress, but can be reduced to first order equations.

In[2]:= compateqs = {

$$r \left(\mathsf{D}[\mathsf{el} \epsilon \Theta \Theta 1[r], \{r, 2\}] + \frac{2}{r} \mathsf{D}[\mathsf{el} \epsilon \Theta \Theta 1[r], r] - \frac{1}{r} \mathsf{D}[\mathsf{el} \epsilon r r 1[r], r] \right) = 0,$$

$$r \left(\mathsf{D}[\mathsf{e2} \epsilon \Theta \Theta 2[r], \{r, 2\}] + \frac{2}{r} \mathsf{D}[\mathsf{e2} \epsilon \Theta \Theta 2[r], r] - \frac{1}{r} \mathsf{D}[\mathsf{e2} \epsilon r r 2[r], r] \right) = 0$$

$$\};$$

The roller is relatively short, and the end faces are traction free, so we will assume plane stress conditions. In that case, the relationships between strains and stresses are:

Hooke's law: Plane stress

$$\epsilon_r = (\sigma_r - \nu \sigma_{\theta})/E, \qquad \epsilon_{\theta} = (\sigma_{\theta} - \nu \sigma_r)/E, \qquad \gamma_{r\theta} = \tau_{r\theta}/G$$

(See http://www.ecs.umass.edu/~arwade/courses/str-mech/polar.pdf) Due to axisymmetry, there is no shear stress.

The stress-strain relations (constitute equations) are:

 $\ln[3] = \operatorname{err1}[r_{-}] := \frac{1}{e1} (\operatorname{\sigmarr1}[r] - \nu 1 \operatorname{\sigma\theta\theta1}[r])$ $\ln[4] = \operatorname{err2}[r_{-}] := \frac{1}{e2} (\operatorname{\sigmarr2}[r] - \nu 2 \operatorname{\sigma\theta\theta2}[r])$ $\ln[5] = \operatorname{e\theta\theta1}[r_{-}] := \frac{1}{e1} (\operatorname{\sigma\theta\theta1}[r] - \nu 1 \operatorname{\sigmarr1}[r])$ $\ln[6] = \operatorname{e\theta\theta2}[r_{-}] := \frac{1}{e2} (\operatorname{\sigma\theta\theta2}[r] - \nu 2 \operatorname{\sigmarr2}[r])$

Substituting strains in terms of stresses in the compatibility equations leads to the following second

order equations:

In[7]:= compateqs // Simplify

```
\operatorname{Out}_{7}=\left\{\left(2+\nu 1\right)\sigma\Theta\Theta1'[r]+r\left(-\nu 1\sigma rr1''[r]+\sigma\Theta\Theta1''[r]\right)=\left(1+2\nu 1\right)\sigma rr1'[r],\right\}
              (2 + v2) \sigma \Theta \Theta 2'[r] + r (-v2 \sigma r r 2''[r] + \sigma \Theta \Theta 2''[r]) = (1 + 2v2) \sigma r r 2'[r]
```

Integrating by parts and recognizing that $\int r \sigma''[r] dr == r \sigma'[r] - \sigma[r]$ to eliminate the second derivatives yields first order differential equations.

```
In[8]:= compat = {
             (1+2\nu 1) \sigma rr1[r] - (2+\nu 1) \sigma \theta \theta 1[r] =
               -v1 (r D[\sigmarr1[r], r] - \sigmarr1[r]) + (r D[\sigma\theta\theta1[r], r] - \sigma\theta\theta1[r]),
             (1+2\nu^2)\sigma rr^2[r] - (2+\nu^2)\sigma \theta \theta^2[r] =
               -v2 (r D[\sigmarr2[r], r] - \sigmarr2[r]) + (r D[\sigma\theta\theta2[r], r] - \sigma\theta\theta2[r])
           }
\operatorname{out}_{[s]} = \left\{ (1+2\nu 1) \sigma rr1[r] - (2+\nu 1) \sigma \theta \theta 1[r] = -\sigma \theta \theta 1[r] - \nu 1 (-\sigma rr1[r] + r \sigma rr1'[r]) + r \sigma \theta \theta 1'[r], \right\}
           (1+2\nu 2) \sigma rr2[r] - (2+\nu 2) \sigma \theta 2[r] = -\sigma \theta 2[r] - \nu 2(-\sigma rr2[r] + r \sigma rr2'[r]) + r \sigma \theta 2'[r]
```

Boundary Conditions

Boundary conditions are such that the bore and OD are traction-free (i.e., radial stress is zero there), and the radial stress is continuous across the rubber-steel interface.

```
In[9]:= bcs1 = {
           \sigma rr1[a] = 0,
           \sigma rr2[c] = 0,
           \sigma rr1[b] = \sigma rr2[b]
         };
```

If there is an interference fit, then the inward displacement of the core at its OD and the outward displacement of the rubber at its ID must add up to the initial radial interference fit when the parts are separate. In the case of the Fairlane wheels, we assume there is no significant interference fit. We use the fact that $\epsilon \theta \theta = u/r$ to obtain the displacement condition for u[b] at the interface between steel and rubber in terms of the stresses via the $\epsilon \theta \theta$ strains and the constitutive equations.

```
ln[10] = bcs2 = (
                b \in \Theta \otimes [b] - b \in \Theta \otimes [b] = \delta
              );
```

```
in[11]:= bcs = Flatten[{bcs1, bcs2}];
```

Solution

Solve the differential equations symbolically subject to the compatibility equations and boundary conditions.

 $ln[12]= ansr = Flatten[DSolve[Flatten[{deqs, compat, bcs}], {\sigmarr1, \sigma\theta\theta1, \sigmarr2, \sigma\theta\theta2}, r]];$

Use the fact that $\epsilon \theta \theta = u/r$ to solve for *u* in terms of the stresses.

 $In[13]:= u1[p_] := p \in \Theta \Theta 1[p]$

 $ln[14]:= u2[p_] := p \in \Theta \Theta 2[p]$

Set up parameters for calculating the specific wheel at issue and for plotting.

 $\ln[15]:= \text{ paramsPstress} = \{ e1 \rightarrow 200. \times 10^9, e2 \rightarrow 6.14 \times 10^6, \nu 1 \rightarrow 0.29, \nu 2 \rightarrow 0.5, \delta \rightarrow 0.000, a \rightarrow 0.0159, b \rightarrow 0.0175, c \rightarrow 0.051, \rho 1 \rightarrow 8000., \rho 2 \rightarrow 1520., \omega \rightarrow 1050 \};$

Checks and Verifications

Check that the radial stress components are continuous, that is, their difference is zero, across the interface at r = b under the plane stress assumption.

```
In[16]:= orr1[b] - orr2[b] /. ansr /. paramsPstress
```

Out[16]= 0.

Note that the hoop ($\sigma\theta\theta$) stresses are discontinuous across the interface.

```
ln[17] = \sigma \Theta \Theta \mathbf{1}[\mathbf{b}] - \sigma \Theta \Theta \mathbf{2}[\mathbf{b}] /. ansr /. paramsPstress
```

Out[17]= 2.41721×10^7

Check that the displacements are continuous across the interface between the rubber and steel under the plane stress assumption.

```
ln[18] = u2[b] - u1[b] - \delta /. ansr /. paramsPstress
```

```
Out[18]= 2.50425 \times 10^{-18}
```

Verify that small strain theory is relevant:

```
ln[19]:= {err2[b], err2[c], e002[c]} /. ansr /. paramsPstress
```

```
Out[19] = \{0.267983, -0.0332404, 0.0664809\}
```

Note that the radial strain near the interface is moderately large, but still less than one-tenth the strain at break listed above for 60A Durometer chloroprene rubber.

Stress Distributions

```
In[20]:= g1 = Plot[Evaluate[{\sigmarr1[r], \sigma\theta\theta1[r]} /. ansr /. paramsPstress], {r, 0.0159, 0.0175}, PlotRange \rightarrow \{\{0.015, 0.051\}, \{-0 \times 10^7, 3 \times 10^7\}\}, Frame \rightarrow True, FrameLabel \rightarrow \{"RADIUS (m)", "STRESS (Pa)", "Plane Stress"\}, PlotLegends \rightarrow \{"\sigmarr", "\sigma\theta\theta"\}];
```

```
 \begin{aligned} & \ln[21] = \text{gla} = \text{Plot}[\text{Evaluate}[\{\sigma rr1[r], \sigma\theta\theta1[r]\} /. \text{ ansr } /. \text{ paramsPstress}], \{r, 0.0159, 0.0175\}, \\ & \text{PlotRange} \rightarrow \left\{\{0.015, 0.051\}, \left\{-0 \times 10^7, 2.5 \times 10^6\right\}\right\}, \text{Frame} \rightarrow \text{True}, \text{FrameLabel} \rightarrow \\ & \{\text{"RADIUS (m)", "STRESS (Pa)", "Plane Stress"}\}, \text{PlotLegends} \rightarrow \{\text{"}\sigma rr", \text{"}\sigma\theta\theta\text{"}\}]; \end{aligned}
```

The hoop stress is the steel is the highest stress by far.

In[24]:= Show[g1, g2]



Zooming in on the rubber stress distribution.



Displacement Distributions

The steel expands (displaces radially) negligibly compared to the rubber.

```
In[26]:= g3 = Plot[Evaluate[{u1[r]} /. ansr /. paramsPstress],
```

```
{r, 0.0159, 0.0175}, PlotRange \rightarrow {{0.015, 0.051}, {0, 0.005}}, Frame \rightarrow True,
FrameLabel \rightarrow {"RADIUS (m)", "RADIAL DISPLACEMENT (m)", "Plane Stress"},
PlotLegends \rightarrow {"Steel Core"}, PlotStyle \rightarrow {Thickness[0.02], Red}];
```

```
In[27]:= g4 = Plot[Evaluate[{u2[r]} /. ansr /. paramsPstress],
```

```
{r, 0.0175, 0.051}, PlotRange \rightarrow {{0.015, 0.051}, {0, 0.005}}, Frame \rightarrow True,
FrameLabel \rightarrow {"RADIUS (m)", "RADIAL DISPLACEMENT (m)", "Plane Stress"},
PlotLegends \rightarrow {"Rubber Roller"}, PlotStyle \rightarrow Black];
```

```
In[28]:= Show[g3, g4]
```



The Effect of Speed

Stresses and displacements vary as the square of the speed.

Remove the speed from the set of parameters to enable plotting versus speed.

```
In[29]:= \text{ paramsPstress2} = \{e1 \rightarrow 200. \times 10^9, e2 \rightarrow 6.14 \times 10^6, \nu 1 \rightarrow 0.29, \nu 2 \rightarrow 0.5, \\ \delta \rightarrow 0.000, a \rightarrow 0.0159, b \rightarrow 0.0175, c \rightarrow 0.051, \rho 1 \rightarrow 8000., \rho 2 \rightarrow 1520.\};
```

Stresses

The radial stress at the interface grows as the square of the speed.

 $\ln[30] = g5 = ParametricPlot\left[\left\{\frac{60}{2\pi}\omega, Evaluate[\sigma rr2[b] /. ansr /. paramsPstress2]\right\}\right\},$ $\{\omega, 0, 1000\}, \text{PlotRange} \rightarrow \{\{0, 10000\}, \{0, 2.5 \times 10^6\}\},\$ Frame \rightarrow True, AspectRatio \rightarrow 1/GoldenRatio, FrameLabel \rightarrow {"SPEED (RPM)", "INTERFACE STRESS (Pa)", "Plane Stress"}, PlotStyle \rightarrow Black] Plane Stress 2.5 × 10⁶ 2.0×10 INTERFACE STRESS (Pa) 1.5×10^{6} Out[30]= 1.0×10^{6} 500 000 0 **•** 0 2000 4000 6000 8000 10000 SPEED (RPM) In[31]:= UnitConvert [1. × 10⁶ Pa, lbf/in²] Out[31]= 145.038 lbf/in²

The maximum radial stress occurs at the interface, and is approximately 300 psi at 9,500 RPM.

```
\ln[32] = g5 = ParametricPlot\left[\left\{\frac{60}{2\pi}\omega, Evaluate\left[\frac{145.038}{10^6}\sigma rr2[b]/.ansr/.paramsPstress2\right]\right\}\right],
            \{\omega, 0, 1000\}, \text{PlotRange} \rightarrow \{\{0, 10000\}, \{0, 350\}\},\
            Frame \rightarrow True, AspectRatio \rightarrow 1/GoldenRatio, FrameLabel \rightarrow
              {"SPEED (RPM)", "INTERFACE STRESS (psi)", "Plane Stress"}, PlotStyle → Black]
                                           Plane Stress
            350
           300
        NTERFACE STRESS (psi)
           250
           200
Out[32]=
            150
            100
            50
              0
                          2000
                                        4000
                                                     6000
                                                                  8000
                                                                               10000
                                          SPEED (RPM)
```

Evaluate the hoop stress in the steel core, which is highest at the inside radius a:



Displacements

The roller grows approximately 6 mm in diameter or 0.25 in at 9,500 RPM.







```
Frame \rightarrow True, AspectRatio \rightarrow 1/GoldenRatio, FrameLabel \rightarrow
```

SPEED (RPM)

{"SPEED (RPM)", "RADIAL DISPLACEMENT (in)", "Plane Stress"}, PlotStyle \rightarrow Red





Conclusions

Radial stresses at the interface are maximum 300 psi at 9,500 RPM as compared to a tensile strength at break of >1723 psi for the chloroprene referenced above. At 9,500 RPM, radial strains in the rubber at the interface are 27% as compared to an elongation at break of >287%. Hoop stresses in the steel are maximum 4,000 psi at 9,500 RPM as compared to 36,000 psi for garden-variety ASTM A36 steel.

This analysis does not at all support the wrapping the specific 60A-Durometer Fairlane rollers with stainless steel "safety" wire. The analysis indicates that stresses, strains, and displacements would not exceed values that the rubber would be expected to withstand, with considerable margin, provided the mechanism providing the rotation itself was robust and controlled.

Limitations

This analysis assumes that the Seals Eastern neoprene parameters are applicable to Fairlance 60A rollers. We would not expect significant differences. This analysis also assumes that the rollers are free of defects, including manufacturing defects or holes drilled for "safety" wires. This analysis is only applicable to the specific geometry, materials, and use conditions modelled. Any use of this analysis is at the users sole risk.